DELAMINATION DETECTION IN A COMPOSITE PLATE USING A DUAL PIEZOELECTRIC TRANSDUCER NETWORK

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Overview of the Proposed Instantaneous Delamination Detection Technique



Wing skin area with stingers

Composite airplane



Delamination

Embedded dual PZT transducer



Research objective

Development of a new methodology that explicitly considers the effects of varying environmental conditions without using any direct comparison with baseline data obtained from Intact conditions.

Uniqueness of the proposed technique

- (1) Robust delamination diagnosis even under varying temperature using an automatically determined damage threshold
- (2) Effective detection of multiple delaminations
- (3) Application to complex geometric structures by applying the proposed mode extraction technique using dual PZTs

Flat skin area







Composite specimen with PZT transducers

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Delamination and temperature have similar effects



Description of a Dual PZT and Signal Notations



Notations of signals obtained from the dual PZTs

$$V_{ij}$$
, *i* and *j*=*r* and *c*

 V_{rr} denotes a response measured by the ring part of the sensing PZT when the ring part of the exciting PZT is activated. Similarly, V_{rc} is measured by the circular part of the sensing PZT when the ring part of the excitation PZT is actuated.



Signals obtained from dual PZTs

Extraction of the A₀ Mode from a Measured Signal

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5

Overall Procedure of the Instantaneous Delamination Detection Technique



(1) Data collection from multiple paths

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Correlation-based Damage Index

$$DI(i,\Omega) = \frac{1}{2} \left(1 - \frac{2}{n_d} \sum_{j}^{n_d/2} corr(A_0(i,\Omega), A_0(j,\Omega))\right)$$

ncy.
$$DI(i) = \frac{1}{N} \sum_{\Omega}^{N} DI(i,\Omega)$$

 Ω is the driving frequency.

 $A_0(i, \Omega)$ or $A_0(j, \Omega)$ is the A_0 mode obtained from the paths *i* and *j* ($1 \le i, j \le 20$)

The path 'j' is selected as the undamaged paths having same angle and spacing with the path i.

'Corr' is the cross correlation.

d is the angle of the path i. ($d = 0^\circ$, 45°, 90°, and 135°)

 n_d is the number of paths of the *d* angle.

 $0 \le DI(i, \Omega) \le 1$

Instantaneous Outlier Analysis

(3) Fit a parametric distribution to the n-1 smallest damage indices and compute a threshold value

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(4) If value of the nth smallest damage index is larger than the threshold value, nth, n+1th ~ Nth damage indices are determined to be damaged.

Impact-induced Delamination Damage on a Composite Specimen

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Experimental Setup for Impact and Temperature Tests

- The dimension of each PZT :
 - * 9 packaged dual PZTs
 - * PSI-5A4E type
- Input signal :

A tone-burst signal with \pm 10 peak-to-peak voltage

- A frequency range 80 kHz to 120 kHz with an increment of 10 kHz
- Sampling rate : 20MS/s
- Power amplifier gain : 5
- Data averaging : 120 times
- Temperature : -10, 20, 50 °C

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Instantaneous Outlier Analysis (Intact)

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Instantaneous Outlier Analysis (Damage 1)

Instantaneous Outlier Analysis (Damages 1, 2 and 3)

Impact-induced Delamination Damage

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High Velocity Impact using a Gas Gun

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Instantaneous Outlier Analysis (Intact)

Wave propagation paths sorted in an ascending order

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Instantaneous Outlier Analysis (Damage 1)

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Instantaneous Outlier Analysis (Damages 1, 2 and 3)

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Wave propagation paths sorted in an ascending order

- An instantaneous delamination detection technique is developed and validated using data obtained from damage states of a flat composite specimen and a specimen with stringers.
- □ The effectiveness of the proposed instantaneous technique is demonstrated explicitly under varying temperature and using structural components with additional structural features such as stringers.
- □ A fundamental Lamb wave mode (A₀ mode) was successfully extracted by the proposed mode extraction technique using a pair of dual PZTs at any desired frequency without any other special tuning.

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Backup

Correlation-based Damage Index

Damage index (D/)

$$DI(i,\Omega) = \frac{1}{2} \left(1 - \frac{2}{n_d} \sum_{j}^{n_d/2} corr(A_0(i,\Omega), A_0(j,\Omega))\right)$$

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22

$$0 \le DI(i, \Omega) \le 1$$

Lamb Wave Propagation Characteristics on a Multilayer Composite Material

